

June 28, 1930

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SPIN the prop once, and your engine takes hold—**S**—with Socony Aviation Gasoline. No sputtering as you "rev" her up. Then, when you give her the gun, the smooth, steady power in Socony is apparent from the start.

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... these and many others take off July 21 to compete in the celebrated American Cirrus \$25,000 ALL-AMERICA FLYING DERBY

Oakland... Ogden... Omaha... Chicago... One plane will come to rest... again at Detroit before the others... One man will win undying fame... All will share the laurels of true achievement... many may profit from the ample prizes... all will give to the future of flying the greatest impetus of contemporary times. The original \$25,000 offered by American Cirrus Engines, Inc. has already grown to \$33,800 with the addition of special and lap prizes. Detroit, Atlanta, Houston, Douglas and Los

Angeles to date have offered cash prizes to winners of respective laps, and other cities are yet to be heard from. Follow the All-America Flying Derby. Support VERMILION ROBERTS. It with your interest, for it is a tracking race of airmanship and plane and motor construction... a magnificent contribution to the development of the business of flying, whose primary purpose is to promote aviation by increasing public interest in flying as a means of transportation and by demonstrating the power, reliability and safety of moderately priced aircraft.



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to date total
\$33,800

including lap prizes
offered by several
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American Cirrus 16-Engine Engine Inc. passed the Bureau of Standards test for approved type certificate with a rating of 100 h.p. at 2300 r.p.m. This engine equipped with a DeSoto Supercharger delivers over 110 h.p.



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Just a step from the Company's stations, Lake Michigan's Sikorsky Amphibians, the "Northern Eagle" is equally at home on open water or on fields. Her landing gear is easily convertible to raise into or submerge in a matter of seconds. With two 400 H. P. Wasp engines she has a cruising speed of 150 miles per hour and she can fly and maneuver on either engine.

Here is a display of remarkable marine power, luxurious comfort and complete adaptability in landing conditions. Sikorsky Amphibians are proving splendid investments for public utilities, transport agencies, business firms and individuals. The reason for this is interesting. Some of them are contained in a recently issued booklet. May we send you a copy?

Sikorsky Aircraft Corporation, Bridgeport, Connecticut, Division of United Aircraft & Transport Corporation.

This "Eagle," appropriately christened the "Northern Eagle," serves its Company and the Company's hundreds of

like the Montreal, 18, the 18000 lbs. from the exceptional value of its "Northern Eagle." See the electric market on top. The valuable guide to plans is illustrated after dark and is stable from an altitude of more than a mile.

hundreds of customers in making surveys of proposed electric transmission lines, power development sites, gas main and coal lands. Further, the ship forms a tremendously important means of quick transportation for officials, men, apparatus and tools to points of emergency. In many cases the plane is loaded on water,



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Mobiloil in another record flight!

Nyrba Lines establish fast South American mail service

The New York, Rio & Buenos Aires Line, (Nyrba) recently set a new annual record of 7 days flying between Miami and Buenos Aires. The fastest steamer used between these points takes 18 days.

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The "double-range" features of Mobiloil Aero Oils—easy starting and prompt oil lubrication at low temperatures—full body and efficient lubrication at high speeds—make them the choice of the world's leading air lines.

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VACUUM OIL COMPANY

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With their dual jobs of engine fly-wheels and air-screws, propellers of balance is of the utmost importance to aircraft performance. Every Hamilton Standard propeller is rigidly tested for balance, both static and dynamic.

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Equal thoroughness marks the tests for dynamic balance, which are done by checking the "track" of each blade through a 180° arc. None but skilled craftsmen qualify for this exacting work . . . and none but perfectly balanced propellers ever qualify for the Hamilton Standard trade-mark.

Correspondence is invited on propeller problems of any sort.

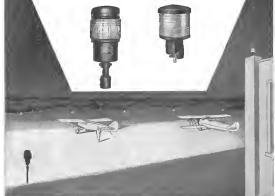
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AVIATION

THE OLDEST AMERICAN AERONAUTICAL MAGAZINE

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EDWARD P. WARNER, Editor

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Covering the News

ON JULY 12th, the first issue of *AVIATION's* new weekly news service will appear. Upon that date, and each Saturday thereafter, there will be presented between separate covers the material for which our readers have been accustomed to look in back of the book, trailing the feature articles. The current news of the industry, of governmental affairs in their aeronautical aspects, and of outside flights and engineering developments is too important to trail anything else. It is too important to be given in any but the briefest and most compact form, easy to find and easy to read. We are giving its standing deserved recognition.

The newspaper, be it daily or weekly, general or specialized, is read and thrown aside—or sometimes preserved for possible needs of future reference. It is read, useful knowledge or entertainment is gained in varying amounts, the effort is pleasurable or not as the case may be, but rarely does the reader stop to figure out just what is being set before him or from what sources. What, precisely, is a newspaper's or news service's task?

Governments of newspaper men, back even before the days of Bennett and Pulitzer, have sped forward in the wake of a shining goldenrod inscribed: "Get the news; all the news." Another and a more exclusively modern group takes as their precursor Mr. Polakoff's Fat Boy, and wants to make yet fatter crop. Another, with Andy Gandy and Mr. Ryan as their standard-bearers, are quite content to amuse. Colaboration rather than exploitation is the measure of their attainment in their own eyes.

Getting the news still affects a problem worthy of a reporter's best talents, and consequent success may be hoped occasionally, but merely filling the columns of a paper presents no problem at all. The difficulty is not so much to acquire news as to reject it. It has not only to be got, but also to be sifted. A great amount of news gets staff. The recognition of the publicity man as a

necessary and a conspicuous member of the staff of every organization has released a flood of material upon the editor's desk. To discriminate between what is of real significance and what is of none, between the proper content of the news columns and material that belongs rather among the advertising, is not a task to be undertaken lightly. It calls at once for all the shrewdness that the editor possesses, for all the knowledge of past events that he can summon to his aid, and for a constant forecasting of the future by way of telling which of today's items is most important to the effect that it will have upon tomorrow. The success of a reporting service such as we furnish, and shall shortly be offering in improved form, is not measured by the number or variety of news printed but by the success with which they tie the past, present, and future together into a logical stream of events. Members of the journalistic craft sometimes cynically remark that there is nothing so dead as today's newspaper, but it is the root from which tomorrow's or next week's will grow.

Any proper presentation of news carries with it the responsibility of interpretation and explanation. Undigested material is simply confusing in proportion to its mass. The average reader has too many troubles of his own to remember, or spontaneously to appreciate, the background of the items that start up at him from the paper. We are willing if one begins to tell him not only what is happening, but what it means. It is as unfair to expect him to recall all the pertinent industrial history as to demand that he depend on his own recollection for the identity of the R12-R3 should dove cryptic symbols appear in the week's reports. All the collateral information, we aim to furnish. That's what an editorial staff is for.

One thing more you are entitled to expect when you buy a newspaper. Appropriate abstracting and indexing

are essential of a proper service. We make it a practice now to pick out the conspicuous occurrences of each week and group them together to provide a three-minute survey of the aeronautical world. We shall continue that practice, but it will be supplemented and extended by other schemes of summary and classification. The headline reader has been subjected to a great deal of (unintended scorn). The busy man reads headlines for want of anything better because he has no time to wade through an ocean of verbiage, but no man is so busy as not to wait at his disposal the unbridled amount of those entries that most deeply interest him. To be at once concise and complete is difficult, but at least the concise form can achieve the complete one. We are anxious especially to render service to the man who wants to get the week's story between telephone calls, but who dreads also that there are two or three chapters in that story which he must have every detail.



Promoting Aviation.—How Not To Do It

EVERYONE connected with aeronautics now agrees that intensive conception of aviation is a poor recommendation for a commercial pilot. Everyone agrees that the arena of Georgia and South is a poor last with which to catch business men in passengers for air transport lines. There is nothing, if not quite a universal consensus, among those actively engaged in aeronautical work and dependent upon the growth of the airplane's popularity for their livelihood, that it is not only a bad thing to drink too much but a bad thing to spend much time in talking about the subject.

We have touched upon this subject from time to time in the past always in general terms. Now we are going to be specific, and call a few names.

There has reached this office an announcement of an aviation meet at Alhambra, Pa., to dedicate a new airport. It bears the heading of the Herbert Paul Lewis Post No. 29 of the American Legion. It proclaims that "Pilot's Party" is planned that might get real enthusiasm. Who knows? And further, in the most conspicuous location on the sheet, that "According to Flying Globes, the Liberty Digest poll shows Alhambra is

1,039 for referendum
2,563 for authorization
5,001 for repeal!"

Emphasis is supplied, beyond the range of our editorial power to reproduce, by printing the first of these three in the most distinctive of types, the last in screening letters a quarter of an inch high.

At the top of the page, we note with interest, stand-

the motto, "For God and Country." At the bottom, "All motorists and agreements, to be legal, must be signed by the Post Commander and Adjutant." This particular proclamation carries the same note of the character of the contest committee of the meet, so the implied promise conveyed by the forwarding of the "Pilot's Party" cannot be considered lightly binding.

If the sponsors of that document want to lay their plans to have a good time among themselves in any way that most appeals to them, we have no comment to offer. If they care to broadcast the news of it to the world, that's up to them. It's a free country. So far as we know, no one has any control over their plans and their advertising unless the national officers of the Legion, who have always shown themselves sympathetic to a sensible development of civil aviation and anxious to assist it might care to try to exercise a little. But if the officers of the Herbert Paul Lewis Post (assuming the announcement to be genuine), or of any other group, think that this is the way to promote flying, or that there are rendering any service to aeronautics with a sheet so published, we can very positively inform them that they are mistaken.

Builders and operators of aircraft have enough natural and inevitable difficulties to contend with at the present time without having their supposed, but certainly badly misapplied, friends laboring to set up new obstructions of unfavourable public opinion in their path.



Aviation's Indirect Service to Industry

ONE OF the subtle points in aeronautical development, often overlooked upon its co-operation extended aviation by industries whose fundamental interests are far from aeronautical, blash of our technical progress has been due to alien industries which concentrated their specialized resources on the development of specialized materials and parts for aeronautical applications and in so doing increased the strength-weight ratio of their products. These who have participated have helped along their own fortunes, for the aircraft business obtained, significant though it has been itself, has been but a small measure of the total reward for their effort. Almost invariably they have been able profitably to apply the progress made, and the technical lessons learned to the far greater volume of the basic markets for their products.

More and more, of industry that is concerned with moving parts is appreciating the significance of moving a certain weight. More than is at a distinct disadvantage. Every pound moved, whether it is traveling on wings or slithering to rest in a mechanism, contributes to cost. The cost of carrying weight are higher in aircraft than elsewhere, but the principle is universal of

June 25, 1918

June 25, 1918

If the same strength is a material cost be attained with a lesser weight, or a given function performed with a lesser weight due to a better load distribution, the gain is self-evident, no matter what the industry. Many of the high-strength steels and light alloys used in general industry, and many of the weight-saving principles of stress analysis applied, owe their origin to aviation.

There are nevertheless, here and there, natural producers or parts makers, who do not see the value of development for aeronautical use. They feel that it is present if aviation wishes to utilize their products,—"Well—there they are, and how many will you have, please?" If, or when, aviation can persuade themselves as they are accustomed to in other industries, they will be glad to consider the production of modified materials or specialized parts. They are a minority. We believe them mistaken in their attitude.

Entirely apart from any aviation volume at present available, they are overlooking the advantage to themselves of securing the active co-operation of the world's most highly skilled experts in weight reduction. They are virtually adding to their research staff, by the mere acts of co-operation, men who have attained the front rank of technical achievement in strength-weight ratio designs. The advice that they get, and the experience that they gain on aircraft will help them enormously in developing materials or designing parts for greater strength at less weight. To all such we extend a cordial invitation to connect themselves actively in the aviation field, to the mutual advantage of aviation and their own present business, wherever it may be.



Advertising Safety

THE OBJECT of good advertising is not primarily to hypnotize the reader. It certainly is not to delude. The first need is that attention should be attracted, but when spectacular display and striking slogans are their attempt to insure that the eye will be caught something specific must be said. Some information must be given. The advertisement of an airplane must be something more than a mere declaration that such a machine exists. It ought to contain a concrete assertion, if possible with detailed evidence, of at least a few specific virtues.

There are many points upon which the builder of an airplane may feel that he has something to boast of its qualities, and most of them find their way into the advertising pages. Examination of a few recent copies of AVIATION gives evidence that manufacturers, or their advertising agencies, are quick to seize on chances of economy in operation or simplicity of maintenance. They are in no wise backward about showing their perfor-

mance figures to become known. They are diligent in compiling and reciting statistics upon the number of mishaps of their own organizations, as compared with those of their rivals, that are in regular service. If they can show a price advantage, they manifest no undue diffidence about mentioning it.

One subject, and only one, seems to be virtually overlooked, or entirely taken,—safety.

It is sometimes mentioned in the abstract. A plane may be "the safest in the world." It may be credited with specific features contributing to safety, although one that is comparatively rare. Almost never,—we have encountered but two major exceptions over the last few months,—is there any attempt at proving just how safe the craft can be. Anybody who is inherent safety as in their other qualities. Some types have records far superior to the general average,—records of which their sponsors have every right to be proud. So have some air transport lines. Why don't they celebrate the fact?

The explosion must come down from the bad old days when so one talked about safety at all, and when to mention the subject in any general gathering was a sort of treason to aviation. The effort upon the inevitable layman is a potential air traveler was of course to convince him that behind an extraordinary conspiracy of silence there lay some very ugly secrets, and that we did not mention safety because we did not dare to have the truth known.

The counter-argument has been that if we did not talk in too much detail about safety the traveling industry would also eliminate the topic from their minds. That doctrine was espoused by some of our most distinguished operators, but we have always believed it essentially unsound. It is with gratification that we have seen the subject of safety become an open one, its importance largely recognized, thanks largely to the initiative taken and the frankness displayed by such leaders as Basil Chambers and Paul Henderson.

But now there is another step, not merely to talk safety among ourselves but to advertise it to the public and in the most specific terms. Aeronautics have been advised with safety as their pre-adjacent selling point. Railroads with splendid operating histories have proudly proclaimed their safety record to the public. It is true that the aviation industry followed suit, on a grand scale, not merely in a few specific instances.

There are airplanes which have never had a structural failure. Why not proclaim the fact? There are others which have never had a fatality resulting from an involuntary spin. Why not say so? There are air lines which have never had a serious accident. Why not let the public in on the secret? It is just a pleasant thing to talk about accidents even for the purpose of emphasizing their absence, but the potential purchaser will think about them in spite of all we can do. It is better to reassure him specifically than to pretend that he cannot feel any reassurance.

THIS MATTER OF SALESMANSHIP AND THE Truth

By ROBERT M. BURT

WHY do people not buy planes and fly themselves? Or if they do not care to learn to fly, why do they not purchase a plane and hire a pilot? Various reasons have been advanced as to why they have not. Lack of service facilities at airports, inadequate number of flying fields, general ignorance about flying and its advantages among the public at large. Some hold out that the general public is in a state of apathy regarding flying. It is said they are interested in aviation as spectators or in an impersonal way, but do not accept it for themselves personally. Some claim the reason is poor salesmanship in selling planes and flying courses that the cost of aircraft and learning to fly must come down. Many other reasons have been advanced as to explanations. A thorough investigation by the author convinces him that the real reason is none of these enumerated. They are contributing factors, it is true, but none of them is the fundamental reason. The aviation industry has been kidding itself about this subject long enough, not alone kidding themselves, but trying to kid the public and thinking that they could get away with it. The cause of the hydro-headed monster in flying airmen or to put it in another way, it is lack of flying skill. Characteristics of the aviation industry in the past two years whenever that ugly specter protruded itself as the industry's line of vision they have laughed and

and, "out of my sight, you, we have too many other things to think about now. We'll take care of you later." The time has come when the very life blood of the industry is dependent upon their no longer ignoring necessary conditions or obstacles. Instead of sitting around among themselves congratulating each other on having gotten in on the ground floor of a marvelous industry, they are going to have to sit down many long weary hours, face their katechisms honestly without ever looking a single one, understand what they have to contend with, and then struggle for solutions. If they do not do this there won't be any industry of which to be so proud.

Most aircraft salesmen tell people that flying is just as easy as driving a car. "The thing is a crash—just like rolling off a log." Such claims are entirely wrong and misleading, and by going out with such stories the industry is actively going to get, and is getting, in wrong with the public. If it keeps it up long enough it will lose the public's confidence. The public will back aviation and support it provided they are told the truth. There is nothing in the truth of which to be ashamed. It is true that aviation is many respects as odd in the experimental stage. There are many things yet to be learned. That there is no reason

whatever why the public should not be completely taken into the industry's confidence in all things pertaining to flying.

If what you say regarding the amount of flying skill necessary is true how much does it take, you ask? Very well, suppose we do a little investigating. The easiest licensing to obtain, the one which requires the least amount of skill, is the Private Pilot's License. Before starting flying a physical examination must be passed which, though not very severe, could not be passed by everyone. It assumes absence of organic disease, fairly good vision and normal judgment of distance. The applicant must have had 10 hours of solo flying although if he is a graduate of an approved civilian school, part of his dual instruction time may be credited as part of the 10 hours. There is flight test must be taken which includes a series of five goals and three steep figure eight turns from 800 to 1,000 ft., respectively. A spiral in one direction from 2,000 ft., with engine throttled and hand as normal landing attitude by which touching ground in front of and within 500 ft. of a line designated by an examiner for the Department of Commerce. In addition three arbitrary landings must be made before the Inspector. The applicant for the private license must also pass a written examination on the air traffic rules and those portions of the air commerce regulations pertaining to pilot's privileges and limitations and to the inspection and operation of aircraft. Most reputable schools charge between \$500 and \$600 for the course leading to the Private License. Average students will find that by the time they take the examination they will have had a total of around 23 hours in the air. The length of time required to complete such a course and take all the examinations will vary depending upon the ability of the student, general weather conditions and the particular season of the year. It can be done in good weather in a month, provided the student can devote a very large amount of time to it.

How much skill does it take to drive an automobile? It would be a waste of good time to discuss this subject. Every motorist knows how much time it takes to learn to drive a car. Some people pick it up in a few minutes others may take several hours. As for expense—there is none. If anyone wishes to purchase a car and does not know how to drive one there is no question in the author's mind but that the prospective purchaser will find a motor car agent who will be glad to teach him how to drive in order to make the sale.

After a private license is obtained and the ability to fly has been acquired, a beginning only has been made. In order for the pilot to be able in his handling of a

There is nothing that will cause the customer to fly any quicker than an unconvincing misrepresentation of the merits of a product. For some reason many salesmen in the aircraft industry consider the average plane prospect to be deaf, dumb and blind, particularly dumb. And being that way, he will swallow every bit of the salesman's patter. In this article Mr. Burt deals with such a situation and points out in a most interesting way how it is hindering sales today, and if allowed to continue will practically eliminate the private plane market.

plane he or she must fly regularly. How much, you ask? Opinions differ on this subject but after asking such men as Department of Commerce inspectors, Chief Pilots, captains, pilots of transport planes and a few private pilots an average of all of these opinions would be about two hours a week in which would be included five landings and one emergency landing without the use of the engine, stretching the minimum of the average landing. In other words this is the average amount of what it takes for a pilot to keep on flying sharp.

Old experienced pilots recognize the necessity of doing this. They realize that they must be prepared and be ready to take care of the emergency situations that is bound to crop up every so often in air navigation. Such a situation being usually due to either engine failure or bad weather conditions. Not long ago the author asked the following question of the operations manager of a large air transportation line which has a large number of experienced pilots working under him: "How many of your pilots would be able, from an altitude of 1,000 ft., to fly over average country, to land his plane without injury whatever to the plane, in case of a complete engine failure?"

He thought for a few minutes and then replied, "This subject has troubled me considerably. My pilots do not get enough practice in forced landings. We have good engines, carefully cared for and such occurrences have been almost unknown with us. However, I am requiring my pilots to take a plane up over the field at least once a week at approximately that altitude and simulate such a condition. In answer to your question I would say, at the present time, that probably about 25 per cent of my pilots would be able to fulfill the condition you outlined. Nobody would get hurt probably but in a fourth of the cases probably minor damage would be done to the plane."

Best of all, these men are all experienced transport pilots, the kind in the business. If they cannot be relied upon 100 per cent to get a plane down without damage in forced landings occurring from such an altitude what can be expected of private pilots?

Whether or not the owner of an airplane, flying it himself, could get it down all right under such circum-



Part of the many thousands that visited the 1930 National Air Show at Cleveland, Ohio

cannot depend again to a great extent on the type of plane he is flying. There is a vast difference, as everyone knows, between the performance of some of the experimental gliders offered to the public at the present time. It is ridiculous to compare an open cockpit training-type plane with a very slow landing speed and an ultra-powerful cabin jet capable of a high speed of over 150 m.p.h., and landing under 25 m.p.h. Anyone can get into a glider, but the vast difference in degree of skill required in handling two such opposed types is a forced landing.

In order to investigate this subject more thoroughly the author interviewed a number of ex-pilots active. At one time they had performed a stunt but did not again at now. In some cases where they had received licenses these had been allowed to lapse. "You purchased a plane at one time and flew it, I told?" "Why do you not ask me now?" What made you part company with it? "Was it because you could not obtain adequate service facilities? Or was it because you had to leave it out in the open?" Or was it too expensive to keep it?" Perhaps you finally got tired of flying? The answers were various. "That is to say they were all flavored differently but in essence they were the same. One man said, 'It is not one of the reasons I was flying. I was flying every other day and got into a bad state. I tried to land in various fields but the wind and weather conditions were so bad that although I tried I gave it up because I did not believe I could get the plane down without crashing up. I thought I was far away from the storm as I could and landed. I just came to the conclusion that there was something lacking in me or it took too much skill to handle a plane as I told it'."

ANOTHER said, "I flew regularly for a while but although I had done very well at flying school, when I had my engine quit one day I finally succeeded in getting down into a field, only after making one break and a jump and two wing patches. It was not the engine I mishandled but I did feel a tremendous fear. I was very nervous and I did not have enough time to do very much of it. I could not keep my skill in handling a plane polished up to the necessary degree to be sure of myself in such a situation. It is true engines are getting much better all the time but they are not reliable and sometimes that or other such an occurrence will take place and I doubted my ability to be able to put the plane down safely every time such a thing happened."

And so it went on. They all had different stories to tell but there was only one common to them. There was some way at least the average, if not a little above the average, of potential private pilots. They felt from personal experience that there was too much skill involved. "Well, what's going to happen you ask—a flying school all just for the fortunate few? For only those who deserve their license to it and make it their profession? Won't the average citizen fly?" Certainly they will, but let's face the situation squarely. There are many things that will help. Adequate airports in great numbers, conveniently placed so that they will be readily accessible to the cross-country voyager. There will have to be more airports in or near the larger cities. They will have to be more conveniently placed and more easily accessible. The private pilot will not do much flying from the large commercial airports or air transportation terminals. We are actually facing the time in the near future when Department of Commerce regulations will prohibit it. They will be forced to fly by the books of

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an passenger and mail lines and it will be right. It is ridiculous to endanger the lives of air passengers and air mail pilots by having the skies shared with airports crowded by unskilled pilots.

It is interesting that many in the public concerning the slight degree of skill required to fly, many airplanes have believed thus the impression that they are difficult to command or run, therefore, refrain. Either that or they have given so serious thought to the subject merely repeating pseudo-fashions, since this idea story learned by heart. Any intelligent thinking man pictures their little balloon, unskilled. These foolish reasoning persons have their story on speed, and speed alone. They will tell you that air accidents were hard to command that the automobile, traveling at the dangerous rate of 20 to 30 m.p.h. was safe. They were hard to command, they will tell you, because, air accidents had been used in the horse-drawn vehicle, which traveled at the safe speed of 4 or 5 m.p.h. They declare that the present provision, which has come to accept automobile speeds of 40 and 50 m.p.h. are safe, are tough to convince that airplane speeds of 100 to 120 m.p.h. are safe. However, they will be able to do it, they say, just as they accomplished the past of the present transition. These arguments would be perfectly sound were it true that the airplane traveled in the same plane as do horse-drawn and motor vehicles. Unfortunately, this is not true. There are some radical differences of which the thinking man is well aware. The first two classes of vehicles go forward, backward, to the left or right, in the same plane. The flying machine, on the contrary, goes forward, to either side, up or down, and, curiously enough it cannot go backward under its own power. It operates, then, in three dimensions, whereas the former two only operate in two dimensions. Don't let these differences lead you that this doesn't make SOME difference. Another factor of extreme importance, in the case of the horse-drawn and motor vehicles, extremely high speeds are dangerous. On the contrary, speed to an airplane means safety. The faster the air passes over the controlling surfaces the more control the operator has over his plane. In lack of speed only is there danger. Don't think for a minute that this second fundamental difference is easy for people to grasp. They may comprehend it theoretically but it takes hard work and experience to put it into practice. In the nature of these basic distinctions let's have done with baloons and tell the truth.

At a recent airport show, while wandering around the floor, I became interested in a rule talk that was being made in one of the booths. I heard several in the offing pretending to look at the plane but actually standing in a lot the salesman was stating. He showed the prospect all over the plane, told what it would do and I must caution that after listening to his presentation the only way that I could visualize the plane he was talking about was to close my eyes, imagine the high speed performance of an army pursuit plane, the carrying capacity of a transport and the landing speed of a glider. Then when I opened my eyes and looked at the plane he was talking about the reaction was pitiful.

The salesman was making a desperate effort to close the deal. Finally he said, "If you will buy this plane and give me a deposit today I'll throw in a \$500 flying school free. You can come in the time in the future, learn on our own field and in two days you will be flying the job back home."

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The prospect made a loud of a joking reply, stating that there must be a lot of profit in airplanes or they would not afford to do that. His eyes were in both thoughts. There was not any profit in the plane and they certainly could not afford to teach him to fly free, that is if they did a thorough job. The prospect asked how much time it would take for him to learn. The salesman replied that he would have him flying in five hours and then he would be able to fly by himself.

"Do you mean to say that I can learn to fly in five hours, take delivery of the plane and it will be perfectly safe for me to fly the plane back home and take my friends up?" If that is the rate for doing instruction is higher than I thought. Went a \$100 on the spot?

THE SALESMAN winked a little but at the end said that it would be a good thing for him to practice a bit before he started back home. I could see nothing enough that the prospect did not believe the salesman. This was unaccounted. There is no question in any mind but that such misleading information about airplanes and flying is doing an immense amount of harm to the industry at large. I have heard intelligent business men say that they believe aviation is just a gossamer proposition unless you can hear your car at I have just reported, in all sincerity, you they be blamed?

Interested to see what this particular prospect would do I followed him when he left the particular booth. If I had been in his shoes I believe that I would have left the building and forgotten all about airplanes and flying, but besides having a glutton for punishment he was honestly interested in securing some information. He finally wandered to another plane and while examining it was taken in hand by another salesman. This sales representative pretended to buy the house facts regarding the size of planes he was selling. He did not con-

template their performance. He gave him an unusually concise and clear mental of figures having to do with the question came of his line of planes. He stressed the all-around efficiency of his planes. They were not so high speed as to cut into their slow landing qualities and their carrying capacity had been stretched to a point that would interfere with safe maneuverability at all times. He told the man frankly that he could not expect to have the minimum of every desirable quality all rolled into one. It was plausibly impossible. In answer to the prospect's question about learning to fly he replied:

"I presume you desire a car. Well, flying is much more difficult than that. However, if you have good eye sight, a good heart and your physical reflexes are normal you can learn to fly. It will be perfectly safe, after having gone through a sufficient course, for you to fly any of our planes where it might not be a plane of skin-high performance. We have tried to build into our line of merchandise qualities which will fit the ability of the average person who might learn to fly. We do not teach flying ourselves, however, so we recommend a good school where you can get a course leading to a private pilot's license for \$150. Even after you get your license you should practice a lot under proper supervision and keep in shape by flying constantly."

This salesman I could see, made a thorough good impression. He treated his prospect as an intelligent business man and did not try to blind him with a bunch of outlandish baloons. If this prospect bought anything I am willing to wager that he at least came back and talked to the salesman who, I am sure, had won his confidence as he had mine. And I, let me add, was much more correct than the prospect. How much better all the aviation industry will be when they have this type of personnel promoting its entry to the public.



Future airplane prospects find an instructor interested. Mr. Henry W. E. Parker, instructor.



By G. M. TURNER

1st Lt. John Turner
Oakland Civil Municipal Airport

PERHAPS the most pleasant feeling experienced by an airport manager is his observation crowded they come when he gives and is greeted by them and flying people as they set down on his field. He must care for their needs and comfort, see that their planes are adequately serviced and perform a multitude of other tasks attendant to be met. But it's all in the day's work. Service is all that any airport has to sell and service is what counts.

I would say that personality, congeniality and courtesy are the three primary requisites of an airport manager—without, of course, overlooking the important factor of ability and dependability. The evils of the ignorant and the stiffness have no place on the flying field. Smile with them and work with them—and they'll all come back again. That, at least, is our motto at Oakland Municipal Airport, and we feel that we have earned a degree of success following it out.

To become an efficient airport manager, one has to combine executive capacity along with optimism and sometimes courage. He must particularly be a dreamer for discipline, for without discipline at an airport there, property and morale are jeopardized. As in a military organization, there has to be respect for authority. I do not mean to say that civil airports should be conducted under military rule, but I do say that airport managers must be clothed by their states and communities with sufficient authority, under reasonable rules and regulations, to get quick and efficient action without question.

In my opinion, those in authority at airports are not delegated sufficient authority to act in coloring field rules and regulations. If he must be a martinet under certain conditions, then the airport manager should be looked to the utmost by fullest authority, that is, of course, if the authorities have confidence in his knowl-

edge, experience and judgment. If they haven't they should get rid of him quickly.

In employing an airport staff, as in any other organization, men should be carefully selected for their capability and experience. They must be trained for their jobs, for they cause most many emergency problems requiring men and staff. The airport staff must be sufficiently large to permit smooth and efficient operation in all departments. Too many mistakes are being made by too few employees. On the other hand, it is true also that too many employees mean cumbersome operation and a drain upon resources.

A reasonable rate should be charged those using an airport, and my experience is that the flat rate is most desirable. I would have no fee for landings, but there should be a reasonable rate for longer storage. Other charges for commercial privileges will help pay for equipment and perhaps bring a return on investment. Commercial privileges include schedule and long flying, student instruction and aircraft demonstrations.

Much difficulty is being experienced at certain airports in the handling of individual operators, that is, the small plane operator. In his own organization, he may be the proud possessor of one, two or three planes and employ his own mechanics and pay rule makers, but to the airport manager he is probably only one of several that have to be treated alike. If the operators are competent, their ability must be kept from getting so soon that their conduct and the conduct of their crew shall be on becoming and useful. The airport manager is frequently called upon to settle disputes among them, and again he must exercise tact and fairness.

UNDERSTANDING the greatest difficulty an airport manager has to face is the fact that innocent fliers just arriving are not schooled in the rules and regulations of his field. Many times they cause considerable traffic congestion. At the department at least, this congestion can be avoided by appearing the fact of the inherent rules. It is wise also to give him the necessary information about landing, for he will probably come again.

Of course, the great trouble so far has been that there has been no standard set of rules for the airport manager that each airport has customs that are peculiar to itself. Nearly all airports have their own rules and regulations. A simple example is the case of the airport that requires left hand turns on takeoff and landing as against that

PROBLEMS OF THE Airport Manager

A Short Discourse on Some of the

Conditions He Must Face, and the Way He Should Fill His Job

which requires right hand turns. It would be well for a standard to be established, though it will be necessary to make it flexible enough to apply to the different conditions obtaining at different airports.

Still another problem is the providing of adequate hangar space. Most of the major airports are over-crowded, though many of them are fast constructing more and more hangars. One of the great hazards of crowded hangars is the difficulty of removing planes quickly in case of fire. Short space and the proper placing of planes is the only solution.

THE MISCONDUCT factor who cannot be overlooked by an airport manager, for upon the ease and dispatch with which a person may reach his field may depend the popularity and success of the field. He should make every effort to see that adequate transportation facilities to and from town are available. He will generally find the street car, bus and taxi people open to suggestions about service, and he will always find the local automobile

association willing to work with him in solving the traffic problems on the road approaches.

I have mentioned standard rules and regulations in field flying operations. Too much emphasis cannot be placed on this need for some kind of standard. I have found that office standards are also desirable, that is, standards as applied to systems of accounting and record keeping. My experience has shown that at least 25 different forms are required for recording routine work. Trial and error have shown me many short cuts, and I will be glad to share my findings with any other airport managers.

Too much emphasis cannot be placed upon the necessity of keeping an airport free from political or "insignificant" interference. Again it must be remembered that safety of lives and property are involved in proper management. Authority should be given only where it makes efficiency. That is almost too obvious to require mention, but many an airport is still struggling under such unnecessary hindrances.



An aerial view of the Oakland Municipal Airport, Oakland, Calif.

Organized AIRCRAFT REPAIRING



A bent rib prepared for removal of beam to rib

ORGANIZED aircraft repairing requires a precise classification of work in order that each man for service may reach the proper department with a minimum of delay. Usually such matters may be attended to during downing or transportation.

Small repair depots may desire a division of rebuilding into three major sections, to be subdivided as the directions increase. These divisions are: Power unit section (including engine, engine controls, instruments and accessories, tanks and lines); Wood structure section (including wings, fuselage, coverings and rigging); Metal structure section (including landing gear, engine, control system and fittings).

Time required for engine overhaul and repair is controlled almost entirely by the capability of the men who do the work. Special tools and an orderly procedure in dismantling and assembly, with much available in the controlling factors. In general it is advisable to remove engine and clean for ordinary top overhaul and repairs, because in the process important fuel connections, belts and fittings come under the direct attention of the engine expert.

Time may vary in contrast with efficient service, yet there is a saving of time in the washing of tanks and lines, checking of fire extinguishing apparatus effected thereby. Further, an accurate knowledge of vital factors in the whole power system where removal is preferable to simply replacement while there must be a choice. It is generally wise that in the average plane damage, the engine is ready for replacement before the plane has been repaired. Hence the purpose here is to outline methods and special equipment to facilitate the work and the quality of work in damaged planes.

Wings usually absorb the major portion of repair time. Yet with proper equipment, one man may accurately rebuild a motor badly damaged (used in 10 hours).

Where it is necessary to replace a solid beam of rectangular cross-section, the entire covering must of course be removed. A wide cut, with a deep groove ground into the middle of the cutting edge, of sufficient width to accommodate the greatest portion of wing strip will be of assistance in cutting the rib loose. In some instances a slender blunt pointed double edged knife will assist as well in releasing the time required for this operation.

With the covering removed, all nails that attach portions of the rib to the damaged beam must be removed. This is usually done with "Magnum" cutting pliers. The

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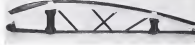
Some Interesting Tips on the Best

bottom cap strip is released, loose gussets immediately adjust the beam by drawing web and separating at the glued joint with a thin, sharpened "pelican" knife. A light tap with a mallet will usually release angle sections attached to the beam and a thin hand-saw blade run through to move against compressed nails and in some clearance for passage of beam through the ribs. Various air brace wires is next released, and all bolts through the beam completely removed, as well as locking blocks, braces, curved wing tip and brackets holding webbing or butt rib.

The cross bracing ribs to removing beam, a "U" section strip slightly wider than beam thickness, and of proper length, with paired gussets secured to one end the rib, as required. A double "transom" jig attached to one end of this long jig, with screw adjustment to force the web into the side of the beam will permit "cracking" the beam out, without damage to the ribs. The same jig is used in placing the new beam, but with "projecting fingers" on opposite side of rib and the gussets free.

If solid spars are made up in the shop, which is usually less expensive to the plane owner, it is necessary that dimensions be precise in all respects, and drawings should be available. If they are not, a tender cardboard beam may be used as a jig, for having of trans and other fitting holes. Beam material should be of select spruce, averaging from 10 to 15 years to each side, these grains being in nearly parallel to upper and lower edges as possible. The ribs of smaller (red) to summer glaze (white) grains, should be over 1 in. 3, and at least 98 per cent of the grain at one end of the beam should be present in the other. Crowned cabinet spruce is preferable to red or brown, and flexibility may be present on the sides. Sap pockets, if any and wet decay, and slight irregularities of grain are permissible, if these conditions do not come at points of maximum stresses, or infection points of the spar. Scarce and wood checks render such material unfit for spars. Material of a white color rather should be avoided in general, it is either unhealthily wood, or over latent and brittle.

After a thorough examination of the new beams, it is worked through the ribs into position. Where compression ribs have been damaged, the first consideration is to replace these with properly dimensioned members. The next operation is the attachment of drag trans (fir-



A rib of wood is prepared ready for the replacement of the upper cover web

Methods to be Employed in the Correct Repair of Damaged Aircraft

bing and rigging for alignment and position. Certain panels have a wood angle of "half" or "weep" block, which must be handled at this point. Next the ribs are properly spaced, and upright members glued and nailed to the beam, capstrips replaced and ribs permanently assembled as before, checking leading or trailing edge at each rib. For beams alone, replacement ribs should not exceed three lines.

In the of the true type, it is often more advisable to rebuild than to replace with new from a standpoint of increasing time in the work. Short members of a true system rib may be replaced by the removal of one gusset, or if the capstrip is to be replaced with new, they may be replaced before attaching the new strip. It must be borne in mind that the finished work should be identical with the original, and stitching of members over beams, butt joints, side lapping, remodeling of damaged members by "doubling" while others found, it is good practice. A damaged capstrip should be completely removed by means of the true hand saw attachment for electric drill, and replaced with a new one. When making of drag trans, one hour per quarter inch cross-sectional area at the most already been proven will handle the average case section. The toy nails may be handled by means of a magnetic strip rubbing machine, especially designed for this work. A correctly designed trans type rib may be released correctly in about 20 min.

New trans ribs are partially assembled for stock, without bottom cap strip, final assembly is accomplished on the panel, with corner jigs clamped to beams at a point adjusting the new rib. Rib of solid material with lightening holes are in three sections, nose, interior and trailing edge. They are attached solidly to spars, and a cap strip applied top and bottom. Repair of the ribs, if complete is usually a cumbersome, lengthy affair, and it is preferable to replace the damaged sections with new. Cap strips are released and drag trans wires or trans disconnected for necessary replacement. Center drag by center jig is in trans type ribs.

Ribs of solid oval strips, round capstrips and a built-up trans, must be examined carefully for grain separation at glued ends, as these ribs depend upon solid material and carefully fitted and glued wires for their weight strength ribs. Assembly of upper and lower outer ribs, as possible in a single finished jig, with spacers to represent beams, in groups of four to six. Final assembly of repair ribs is on the right beam,

with partially assembled trans ribs, and positively fitted beam gussets.

Trailing edge alignment or any built-up rib is accomplished by releasing top gusset, changing a straight edge of light metal to butt rib and the reinforcing rib adjoining, aligning, and gluing and resoling the gusset. Ordinarily it is advisable to replace these gussets with new ones, as condensation moisture accumulates in the joint while the plane is at rest, and in time the lost material and nails will increase, resulting in the additional drag of an irregular trailing edge.

Aligned with ribs spot be examined for seeping and glue separation. It is not advisable to attempt a reinforcement of such ribs, because in so doing, the glue is placed in tension, rather than shear and does not hold well. New sections will be preferable to the through bolt method, which decreases the strength weight ratio and with proper equipment, the time required is but slightly in favor of the repair.

Metal ribs of the trans type should be braced, if trans members become separated unless the construction is perfect. In this case, reinforcing plates should be applied, as a sleeve, and re-stayed using the original punched holes where possible. Metal ribs formed from the material may often respond in slight corrections of alignment, but a comparatively slight "back" renders ribs slightly for replacement. They are not expensive, and are quickly available in a role.

Built up box spars require precise calculations, and the most careful workmanship. Unless correct blueprints and engineering data are available, repair of this type beam should not be attempted. Double top upper of plywood webs, and "singed" after of compression clamp, not exceeding two feet ahead of wing tip may be safely handled, also reinforcement of thin "wing attachment" half built, by means of a light fitting, allow webs, box cap over damaged end, engaging web behind drag trans fitting bolts.

Internal wing tips are simple in construction, and do not require form, or structure. The fin strips are arranged for grain relation, fresh cross glue applied, and one end of the group clamped to either leading edge or external hinge support beam. It is then worked upward by hand to the opposite side and clamped temporarily. Starting at the leading edge, small clamps at short intervals around the curve are applied, the curve being shaped in these clamps are placed. Unless instructions are to make the shaping should be accomplished with care. As-

assembly should take less than 20 min. and finishing a similar time.

Covering time is reduced by the use of half and full length bars. Wing tip repair usually requires but a single use of material. The bars being joined across the wing, increasing old tape and softening the exposed portion with thin dope, before applying new material. New strip is then "doped on" and allowed to dry, upper and lower surfaces separately. When the new material is permanently in position, it is released in the regular manner starting at the rib spans which join is made, then one cut of dope, tape and finish. Replacement of fabric and repairs may be done inboard with a little care, saving labor on both edges in the manner above. Small patches on fabric should be finished with the same number of cuts of dope as the panel and should be cut to same end shape, diamond, oval, shield or circle.

Even if the speed and accuracy in plane rebuilding depends upon the orderly sequence of operations. In wing work, there are certain fundamental steps that are almost self-evident. It is obviously impossible to "rig" beams for alignment and position until the complete rib truss is in position. And it is useless to attempt rib repair or replacement until beams are permanently adjusted. Further, and ribs must be cared for first to prevent proper alignment of leading and trailing edges.

In the strut section, undamaged bay, cable and inter-strut struts may be used as pins for duplicate members, where no permanent rig is available. Thin hardwood blocks, with countersinks on each side to accommodate the curved sides of streamline tubing, permit clamping the new sections to the old member, in proper position for welding and drilling of bolt holes. Both bolt reinforcements, mostly tube bolings, web washers, or internal clevis, are readily duplicated. Where old struts have been removed, care in aligning threaded portions with the original axes of the tube is essential to avoid aerodynamic loading, and vibration at that point.

An accurate check of fuselage and engine mount alignment can be done without the dismantling of the structure. A graduated gauge bar consisting of fuselage, covering upon a overhead track provides the means. Long adjustable indicators dropping from the cross bar will give accurate measurements. As weight between floor and overhead track, provided with "V" blocks, may which the strongest may be welded to also necessary. It is to be noted that certain places correct engine torque by offset of the engine hub. Slight variations of alignment may also be corrected by fittings. Later work requires jacking, raising or displacing the engine in the usual manner of standard model sleeves covering the ends of original and replaced tubes.

Where a fuselage is seriously damaged, it is well to make a live check of the fuselage structure and engine bases, and run an analysis of forces developed during the contact with the ground during the accident. This is particularly true of cabin floors with rather a complicated truss system. Often a critical load will be developed at a point distant from the most damaged portions that will cause an irreparable crack, or buckling of engine sidewall. Such investigation will usually bring these items to light, and justify the slight effort required.

Any cut, dented or banded stress-tube taking a compressive load is say fabric laminate should be replaced with new material. Building up dents with web, sheet reinforcement and inside blocking will not restore strength, though often resorted to. A diamond shaped

reinforcement, whose width is not to exceed one-third the width of upper end of the surrounding section, may safely be brazed over small dents, material being of the same thickness as the sidewall of tube. However, it is advisable to sand the area where possible.

Certain fuselage fittings may be straightened if not badly bent. This is ordinarily done with the metal cold, as fittings are usually of alloy steel, and will withstand moderate bending without fiber distortion. No set rule may be applied here, however. Fittings of laminated metal require heating to a dark red on the side to be stretched in straightening. Heat treated fittings, which can usually be recognized, or any indicated on the drawings, should be replaced with factory units, unless full engineering data and heat treating equipment is available, with an expert in charge.

Landing gear may only be warranted here. Dented tubes, bent, dented, buckled or torn, invariably require new members. Welds should be examined for cracking prior which often indicates coming trouble. Opening compression tubes may be replaced by saving old repair fitting carefully, and cutting free from around brazed points of valve which can then be removed. These brazed points are smoothed down, and the valve inserted a tube of correct length, diameter, and precise overall dimensions to work freely within the lower member. The valve must be accurately aligned in the upper tube, and the upper fitting is welded on as a cap.

Certain types of gear have hardwood reinforcements within the tube, to increase the release strength. To be of value these reinforcements must extend the complete length of tube. The process is to weld the end member longest first, drive the reinforcing member into position, cut off accurately, and brass upper fitting to place.

Wheel alignment requires releasing the weight from the gear by raising fuselage slightly, applying close heat to portions of tube that will be stretched in straightening, and slowly pulling into position with bar through side hole, slipping upon stabilizer for track, and using automobile wheel spacing jig for track. Occasionally it is advisable to slot a tube, and reinforce with a "through passed" plate.

Smooth work on engine cooling ducts requires a leather or heavy cloth work highly filled with sand, and a medium soft rubber mallet. Cracks, or hot patches are covered with neatly shaped patches, using round head aluminum sheets closely spaced. "Banding" and "braking" are well understood, and done in the usual manner.

If an accumulation, the most desirable work is separated by replacing damaged parts, rather than temporarily repairing them. The most critical scrutiny should be required to locate the required portions inside or outside.

A simple means of "triple check" begins with an orderly sequence of operations. When finished, the main difficulty during the work given all items a preliminary check. Secondary inspection is done by an assistant in training, with inspection card and with a small "free all boxes" for items found. Final check is by both expert and assistant starting at opposite ends and passing.

There are factors of one alone and hidden weaknesses that must be brought to light during the course of the work. Hence, swift accurate and inconspicuous service must be based, not entirely upon efficient production methods, but largely upon special training and knowledge of the subject, the development of particularly adaptable equipment, of which there is no end, and, and a critical turn of mind to those who actually do the work.

USE OF X-RAYS IN THE AERONAUTIC INDUSTRY



Professor Norton working with a typical radiograph set-up for testing of the tube and control in above.

By JOHN T. NORTON
Assistant Professor of Physics,
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X-RAYS have been known for some thirty years, and have been used with increasing success in one technical field, but it is only recently that industrial and material applications have been developed to the point where they may be considered as practical requirements. Up to the present time the actual leaders of aircraft have not employed X-ray methods to any extent, but the important field of applications is in the materials and parts of which aircraft are made.

Because of their extraordinary penetrating power X-rays enable us to study the structure of an object without cutting it apart. This process is known as radiography, and has proved itself very useful. Perhaps it is best described by considering an actual example. Let us suppose an aluminum alloy casting which forms the crank case of an engine engine has been machined by a new process, and it is desired to examine its internal structure. It is sent to the X-ray laboratory, where it is set up to be radiographed. The X-ray tube is mounted on an adjustable stand and it is connected to the high voltage generator. A voltage between 100 to 300 kilovolts is used depending upon the type of work. The tube is set above the portion of the casting to be examined, at a distance of about 30 in., and immediately under the area is placed a piece of photographic film in a thin flat box which protects it from the light. This flat box is moved very slowly with X-rays as with visible light. When the exposure is finished the film X-rays are produced which pass down through the casting, through the cover of the film box or cassette and on to the film. Any defects in the casting will cast shadows in the X-ray beam and thus they will appear on the film when developed. These are very often small and are promptly sharp detail so that defects such as blowholes and inclusions of foreign material are accurately col-

lected in the developed film. Thick samples require a higher voltage on the X-ray tube and a longer exposure than thin ones. The maximum thickness which can be examined depends upon the material and at the present time the greatest limits are aluminum, 8 inches; steel, 3½ inches; and brass, 2½ inches. In this thickness a defect whose area is about 2 per cent of the total thickness can be detected with considerable certainty. In thicker sections the defect is much magnified.

There are several classes of materials in which

Most of the methods of testing materials available to aircraft manufacturers at present involve destruction of the part to be tested. This is not only expensive; it means that the parts which are actually used to build planes can be examined only superficially in most cases. Professor Norton, who has been supervising X-ray research at Massachusetts Institute of Technology for the past few years, summarizes the development of testing metal by X-ray up to the present time, and discusses in detail a number of actual and potential applications of the method in aeronautics.

radiography can be suitably applied, and each has its own particular defects. Castings are probably the most important class from the point of view of X-rays. The greatest objection to their use is any structure in the question of possible discontinuities. Largely for this reason, they are used to a comparatively small extent in aircraft construction outside of the engine itself, but if one could be certain of the soundness of the metal, light alloy castings would be a great deal of use. X-rays have been very successful in examining these and in the development of casting methods which yield a high percentage of sound product. Three kinds of



Dark engine radiograph, horizontal. On the right is the dark radiograph produced resulting from a steel transmission and a metal casting radiograph, vertical. On the left is the X-ray beam as it is subjected to steel. The beam is fixed with a lead foil for X-ray protection.

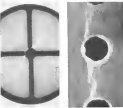
defects are frequently found. Thin holes or inclusions of gas are common, particularly in the sand core material. These may take the form of a fine porosity such as is often found in sand cast coatings and of bubbles which are less frequently in steel castings and the size of these defects can be closely estimated. Inclusions of foreign materials are also common. In steel castings, the sand or a part of the core may be carried by the molten metal to a place where it will eventually reduce the strength. Chaps or core supports and portions of the core itself are often found remaining in the casting, particularly if it has complicated cores, and this condition is also clearly shown in a radiograph. Shrinkage lines and cracks are fairly common, particularly if improper casting methods are employed. Shrinkage lines due to faulty location of gates and risers are easily seen and have a fern-like appearance. Cracks, however, are much more difficult for they involve so much change in the thickness of the metal that unless the X-ray beam passes nearly parallel to the plane of the crack, it will not appear on the negative. Usually, however, a crack is of a winding nature and some portions of it will be visible. Once its presence is suspected, patterns may be taken in several directions, and its extent determined.

With such methods of examination available, many applications at once present themselves. Because of the length of time and the relatively high cost the routine examination of a large number of castings is not a practical procedure, but in aircraft work there are many sections of parts which cannot be allowed to fail in service. Here is a method which gives a complete and

positive view of the interior and yet does not render it unfit for use. Compare it with the only other method of complex examination, which involves cutting up the casting and examining the sections. For it may not have been cut in the place which would have shown the most serious defect and the cutting is needed for actual use. Furthermore, with the X-ray method, the examination may often be carried out before machine work is done, thus saving considerable expense if it proves to be defective. The cost of such an examination although it may even be considerable with that of the object examined, may be repaid in the light of an in-service inspection failure which in aircraft work is usually attended with great serious consequences.

There is another use for X-rays in the examination of castings which is no less important than the one just described. It is a method to be applied to factory practice and consists of the examination of the pilot castings made by the different casting methods. These are numerous factors which have an effect upon the soundness, as for example, the temperature of the metal, the position of the gates and risers, the nature of the mold and the distance of the ladle to the top of the mold as well as many others. Also small changes of composition which do not appreciably alter the physical properties of the metal, sometimes have a great effect on the soundness of the resulting product. X-rays are excellent for examining the samples made by these different methods for it is usually cheaper, easier and more positive than waiting up the casting, which is the only other possibility. In this way it is possible to develop methods which will produce nearly one hundred per cent of sound castings. If one can depend upon soundness the factor of safety or "insurance factor" in the design of the casting need be considerably reduced with the resulting saving of weight, and they would be much more widely used in aircraft construction than at present were this the case.

Welding is used extensively in the building of metal aircraft, and here again we have an example of unusual appearance which is an indication of the interior. X-rays are used a great deal in this type of work, and have proved extremely valuable. Such defects as porosity, inclusions of slag and oxide, lack of fusion between the surfaces, and cracks are detectable. Again we can ex-



Right: Radiograph of the weld in joint at picture. Left: Cracks in metal plate.

amine the product without making it unfit for service. Porosity and other wrought defects are seldom tested by this method because defects in solution occur in this class of materials. However, blades of metal propellers for airplanes have been inspected successfully for cracks.

There has recently been placed on the market a small portable X-ray machine which could be used to advantage for the examination of comparatively transparent parts in the field. The tube and transformer are completely enclosed in a small box so that the danger of a high tension shock is absent and the unit may be operated from an ordinary lamp socket. For inspection purposes, it is possible to observe the X-ray picture directly by means of a fluoroscope although this method does not give as far detail as the photographic method. With such a device, the inspection of an airplane could be made in much the same manner as that of a doctor looking for broken bones. Wound wire spurs and rivets could be given a careful examination without removing the fabric covering. In fact practically all welded parts such as control arms and fittings, valve heads, shock absorbers, tail-rod supports and others could be inspected in place in a very simple yet positive manner. This method has also been successfully applied to weld

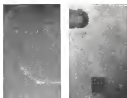


Right: Radiograph of the weld in joint at picture. Left: Cracks in metal plate.

objects in electrical instruments with enclosed wiring, electrically heated spring units and airplane tires.

The radiographic method of inspection has many useful qualities, but there are two important disadvantages which have limited its practical use to a considerable extent. One of these is the cost of the method and the other is complication and lack of portability of the equipment. In most cases it is necessary to bring the objects to the X-ray laboratory where the apparatus is permanently installed, rather than send the equipment to the place where the particular object is in use. These features are being improved, and in the future this method should be very much more useful than at present.

There is another and entirely different use of X-rays which is also of considerable importance in the aeronautical industry. It applies to the study of the fundamental structure of metals and is essentially for the laboratory. A detailed explanation of the method is beyond the scope of this paper, but the physicist has developed it so that it is possible to study the arrangement of atoms in metal crystals and determine the most



Left: Radiograph of weld joint showing in regular cross-section form. Right: Radiograph of weld joint showing in regular cross-section form. Right: Radiograph of weld joint showing in regular cross-section form.

and orientation of the crystal grains in a metal product. This information tells much about the suitability of a material for a particular purpose. Some of the secrets for secrets of the metals are of interest.

It has been found that the type of atomic arrangement in metals has an important bearing on their physical properties. Among the common metals there are three simple types and X-ray methods furnish the only means of determining them in a definite fashion. Pure metals, of course, are seldom used commercially but this study of the real fundamental meaning of the properties which metals exhibit can be extremely to alloys. Thus the properties of an alloy can be accurately predicted.

After a particular metal or alloy has been chosen for a special purpose, its properties can be greatly altered by mechanical or thermal treatment. Consider a piece of properly tempered steel and weld another piece which has been annealed, or a sample of hard drawn aluminum with an annealed sample. The X-ray method gives a more fundamental picture of these changes than the microscope or other metallurgical methods. A particular case is a cold working process such as the rolling of sheet metal. The microscope shows that the crystal grains are elongated in the direction of rolling. The X-ray method shows that these elongated grains are really made up of much smaller grains which are all oriented so that their crystal axes are parallel. Thus giving the so-called definite directional properties. Upon heating this directional effect is difficult to exactly remove although it is completely reestablished later, placed and this has an important bearing on the suitability for further operations such as cupping, spinning, etc.

There are only a few of the many problems of metallurgy which can be studied by this new method, and even though it is not one which the aircraft manufacturer will use to any great extent himself, it has an important bearing on the materials which he chooses for his construction. It is new and comparatively undeveloped, but has many disadvantages which present a very great one. But a great many people are working on its development and rapid progress is being made so that it will eventually become one of our most important means of investigation, for its results are extremely fundamental and since not to be obtained in any other manner.

SOME RECENT DEVELOPMENTS IN *Light Airplanes*

A side view of the
Inland Superport



Inland Sport MONOPLANES

THESE NEW PLANS, being the first ones offered by the Inland Aviation Company. The first of these is powered with a LeRhond 50 engine. The second, a plane powered with a Warner "Super" engine is known as "Superport". Both planes have the same general specifications: The span is 35 ft., the length 19 ft., and the height 7 ft. 2 in. The landing gear is made of welded chrome molybdenum steel. The wings are the conventional type with double tie rods. The landing gear is of the side axle type, using a hydraulic shock

absorber. Some of the features of these planes are: large baggage compartment, demountable engine mounts, and wheel control mechanism. Both planes have the side-by-side dual control arrangement. The sport model weighs 240 lb. empty and carries a useful load of 550 lb. It has a cruising radius of 450 mi. The Superport weighs 280 lb. empty with a useful load of 500 lb. It will cruise 525 mi.

The "Superport" model comes equipped with a complete panel of engine and mechanical instruments. It has emergency lighting, an electric starter

THE Monocoupe "90"

THE "Monocoupe 90" is the second 7500 offering of the Inland Aviation Co. of Wichita, Kan. This plane is powered with a Lambert 8,200 engine, developing 90 hp at 2,375 r.p.m. The Monocoupe provided with this engine has been given Department of Commerce Certificate No. 205.

While similar in general appearance, the Monocoupe 90, differs from its predecessor in the new plane the power has been increased from 35 to 90 hp. The fuselage is 15 in. longer and 4 in. wider than heretofore. The seat cushions are better, staked with kapok, and removable; providing an additional degree of comfort. In back of the seat, is a baggage compartment made of heavy canvas and suitable for important purposes. The entire interior of the cabin is easily traversed. The passenger compartment is 4 in. wider and there are 6 in. more leg room than in the previous model.

An extremely successful extra feature is built in and so arranged that it can be used for a vestibule or has a sliding flap which, if desired, equipment with those on the right hand side being

completely detached. Strides are equipped equipment in all models. A three-way propeller lock is located beneath the instrument panel, allowing the pilot simply to be taken from the main shaft, selectively or both at a time.

The designers have incorporated a novel window which permits the window to be taken out to be open and held in any position while in flight. The entire side seems to look like windows when not in use.

The landing gear is of the split type with telescopic shock struts. The wheels are provided in the lower extremity instead of the upper ones, so in the actual middle of the monoplane. The wheels are the new product of the Aircraft Products Corporation, Detroit, and take the three inch tire size 6.50x10.

Stickler adjustment is obtained by an arrangement of two springs, each variable in its tension through a control inside the pilot's seat. By adjusting the tension the pilot may vary the stick priority. Brakes and Adams are operated by steel rods. The elevator is controlled by a conventional push rod system. The entire design is of steel construction covered with fabric. Two welded aluminum rails, of 15 ft. square each, are built into the wing on either side of the fuselage. These rails are fixed with direct coupling pipes.

The company offers the new Monocoupe in two color combinations, standard standard steel popovers are standard equipment on all models.



The Mercury "Chic" adaptable for training or sport purposes

THE MERCURY "Chic"

DESIGNED PRIMARILY for training, the Mercury "Chic" is adaptable to a wide variety of uses. It is definitely in the light plane class, but has ample steel coupling and tail dual controls. Adjustable seats are provided with provision for the use of parachute if desired.

One of the features of this plane is its all metal construction. The wing spars, landing gear, tail, and fuselage are of chrome molybdenum steel tubing. All other structural parts are of 100% experimental specification steel. The wings and fuselage are fabric covered. The fuselage is fabricated in sections, the control surfaces

in yellow, and the metal parts in black.

The landing gear is of the split axle type and runs standard 20 x 4 wheels. The plane is furnished with a standard LeRhond 50 engine. It has a span of 35 ft. 8 in., a length of 19 ft., and a height of 8 ft. 7 in. The plane weighs 655 lb. empty, and carries a useful load of 520 lb. It has a range of 325 mi. with 20 gal. of gasoline.

While designed primarily for training purposes, the "Chic" is adaptable to use by the private owner. It requires construction along hard nose with two wing supports. The instruments and general equipment of the plane are adequate for cross country flying

"Gee Bee" SPORT OR TRAINING PLANE

THE "Gee Bee," a two place, tail wheel dual control biplane for sport or training purposes has been produced by the Greenville Bros. Aircraft Company of Springfield, Mass. Accordingly, the Gee Bee is conventional. Structurally it is designed to meet the requirements of the Department of Commerce, and much study has been taken in its design to eliminate the possibilities as often found in small airplanes. The design combines maneuverability with stability and will fly "back off" in level or turn flight. It is capable of normal gliding

angle when the engine is feathered. All of the control surfaces on this plane are proportionately large, giving quick response in maneuvers. The designers extend the entire length of the upper wing and center section, with the result that the loads on the stick are not overly high. Over-lift elevator and rudder controls work through carefully fitted leverage hooked levers.

The Gee Bee is equipped with Eadsley brakes and a full wing-brake parachute tail wheel. Adjustment of the brakes is accomplished in a single motion. The main brake lever is shown in



The Gee Bee provided
"Gee Bee" features

the slat of controls is located on the stick assembly and protrudes from beneath the instrument board in the form of a push grip. The brakes are applied by pulling this lever or by pulling this stick back in left hand position. With the levers set, the distribution of braking force between the left and right wheels is automatically controlled by the rudder pedals, thus it is possible to land and stop with the hands and feet at all times on the stick and rudder pedals.

A critical departure from conventional practice is found in the control stick assembly. The sticks, main and end, take the form of push grips, as taken from a door handle, but backrests from beneath the instrument board. These sticks move fore and aft on ball rollers supported by a rubber slide.

The elevator control is translated positively from stick to elevator through a lever and a push-pull cable. Advance control is equally positive, passing from the stick to the cable, the cable extends directly upward to the center section where it is translated through levers and push-pull tubes to the elevator. The use of horizontal sticks gives a perfectly clear cockpit and allows the use of rubber or other stick equipment, aside from the rudder pedals.

The rudder pedals are long from the top and are fitted with flat plates. The rudder pedals—both stick and pedals—may be instantly released by the pilot. It is only necessary for him to turn the handle of the stick to release the stick which becomes completely detached from the rudder. A shift pressure on the pedal adjusts the pilot's rudder pedal is sufficient to cause the stick to drop to their normal position.

Not only does the pilot in student instruction but also greatly facilitates the removal of hands for passenger carrying.

Two throttle levers are provided, one a hand lever with a large handle at the side of the seat, the other a push control in the center of the instrument panel. The lever may be made inoperative by a 90 degree turn of the handle and may be left stationary against the panel.

Two pedals are conspicuously incorporated in the cockpit which is apparently indicated by the use of wheels

in the side splashes. These wells allow the floats and upper arms of the pilot and passenger to pass through the landing water to the lower wing, thus adding the extra weight of the plane.

The landing gear consists of oil hydraulic struts fitted with rubber shock cord. The hydraulic travel is seven inches. The shock cord acts through the last two inches of the travel. The action of the hydraulic strut makes it practically impossible to bounce the plane in landing, while the shock cord gives very smooth landing touches.

All parts of the Desoutter are so assembled that the effects of a crash will be localized to the greatest possible extent. This is accomplished by the use of structural joints in the landing gear struts, and other parts that are liable to damage the primary structure if they collapse.

The wings of the plane have 197 sq ft of area and are made up with a Clark Y section. They have a 53 in. chord, a 20 in. sweep, a 51 in. gap and a 40 in. dihedral. The fuselage is built in entirety of welded aluminum alloy-tubular tubing and covered with fabric. The plane is powered with a Kinner K3 engine. The weight empty is 1,900 lb., 1,500 lb. with a disposable load of 600 lb., and a gross weight of 1,450 lb.



View showing stream lines and location of cockpit on the Desoutter.



The Desoutter three-seater cabin monoplane in flight.

Desoutter CABIN MONOPLANE

THE first British cabin monoplane in the light plane class is the Desoutter three-seater cabin monoplane and is now being standardized by National Flying Services as their first and best machine for general charter. The machine is a high wing monoplane with a wide track undercarriage. The engine is the six-cylinder Heron, four-cylinder version, an engine rated at 165-175 hp, a development from the well-known linked Heron engine of 95-99 hp. Its place has a British Certificate of Approval for a gross weight of 1,600 lb. The top speed is 110-114 mph, and the cruising speed 96-100 mph. Its maximum load is 600 lb. The water which is built into the fuselage is carried in the fuel tanks. The fuel tanks are 1,000 lb. in capacity and are located in the fuselage. The fuel tanks are 1,000 lb. in capacity and are located in the fuselage. The fuel tanks are 1,000 lb. in capacity and are located in the fuselage.

There is no doubt that the high speed is the high speed of the plane. They are, in fact, power, decrease in cost, increase in landing speed and improvement in passenger efficiency. For airplanes, reduction of weight and structural weight are also benefits.

The problem of increasing the high speed by increase of power has already been discussed. The second method, that of decreasing the overall drag of the plane, appears to offer greater possibilities. An examination of modern aircraft indicates numerous features in which the drag of the fuselage may be reduced. Parasites are unstreamlined, are broken in outline by fittings for struts, wing roots, engine cylinders and intake stacks. Open intake, inadequately faired by windshields, are a particularly serious source of drag. Then, too, the wings are installed on the lower portion of the fuselage, immediately behind the propeller, and, thus, in a region of increased drag due to the increased speed of the slip stream. Folds are expected to be working more. They contribute but little to the lift in flight but are a particularly serious source of drag. Any effort designed to reduce the drag of the fuselage or to so construct them that they contribute appreciably to the lift should be of particular advantage. Methods of improving the various features mentioned above will be discussed hereafter.

Racing Seaplanes... PRESENT AND FUTURE



The MacCulloch racer in the Schneider Trophy Race of 1931.

THERE are four recognized methods of reducing the high speed of any plane. They are, in fact, power, decrease in cost, increase in landing speed and improvement in passenger efficiency. For airplanes, reduction of weight and structural weight are also benefits.

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Tests have indicated that a fuselage symmetrical about the outline has but only about sixty per cent of the drag of the average fuselage of unsymmetrical

The Second of a Series of Three Articles—Some Methods of Improving Performance

By JOHN S. KEAN

Project Engineer
Mead Aircraft Factory

cross section. It is apparent, therefore, that while this figure of 40 per cent less drag is not to be expected in the full size plane, considerable improvement may be expected if reasonable efforts are made to attain symmetry.

It is true that the shape of the fuselage is largely determined by the shape of the engine so that improvement in the fuselage form may not be expected until engines are built within a fairly smooth outline. This requirement especially favors the use of four banks of cylinders arranged in the form of an X, or a cross, or of an engine of circular or oval cross section with the propeller axis at approximately the center of area. With any engine of this type a fuselage of approximately symmetrical section is not impossible.

HAVING DECEASED upon the most desirable shape of the engine, the next effort should be directed toward reducing the cross-sectional area of the engine. There is one readily apparent way of doing this, namely, by raising the compression ratio to as high a figure as possible. But there is a limit to the degree to which the compression ratio may be increased and this figure appears to have been reached in the case of the engines for the 5-6 which have a compression ratio reported to be 30 to 1. In America the greatest accident in the use of these high compression ratios have not been advised. However, 30 to 1 appears at this time to be the highest limit for this ratio. Since progress is halted in this direction, the other normal means of increasing the power, that of increasing the displacement, remains. The increased displacement usually requires larger cylinders and pistons with increased bore and stroke and generally increased overall size and weight. Hence, with the normal four stroke cycle engine any attempt to reduce the size of the engine while increasing the power is blocked by limitation inherent in the Otto cycle type engine.

However, there are other types of engines which have not as yet been adapted to airplanes but which seem to present interesting possibilities.

The first of these is the engine employing the Diesel cycle. The success of the Packard air-cooled Diesel engine is well known to the aeronautical world. It would appear that for the same cross-sectional area (or displacement) the Diesel engine could, by virtue of its higher compression ratio, be developed to provide considerably greater power than the Otto cycle engine and furthermore, the Diesel burning use of the heavier grades of oil rather than gasoline, is free from the normal fire hazards and has a much lower fuel consumption factor than the Otto cycle engine. It is also free from the vibration difficulties normally inherent in the standard type engine. Whether the Diesel can be developed in the lean bulk or crawler form, mentioned before as ideal for racing plane and adopted for water or chemical cooling, remains to be seen but such development as of such nature as to seem a desirable contribution to the field of aeronautical power plants. The second type of engine is the two stroke cycle engine. This is particularly a low speed engine compared to a four stroke cycle engine of the same power and displacement. Conversely for the same power and speed, a would seem possible to reduce the displacement and hence the overall size of the engine over that of the four stroke cycle engine. This reduction of overall size may also result in reduction of weight. It is noted that the two stroke cycle engine is not particularly efficient from the standpoint of fuel economy but it is felt that with proper design this drawback may be largely overcome. In any case the two cycle engine has promise of sufficient advantages for a more plane at least, to warrant its further development.

Other changes in the conventional types of engine include the use of the diesel or steam cooling. The former would seem to be preferable to in such case there is no limitation regarding the location of the radiator. Engines should be designed to operate at 300 deg F coolant outlet temperature. This figure is sufficiently above the boiling point of the coolant, ethylene glycol, to allow a margin of safety to prevent boiling of the coolant and is sufficiently high to afford a reasonable air-cooled or steam cooling. The temperature differential to permit satisfactory heat dissipation. With a system so designed the required radiator area may be much reduced over that of a typical water cooled engine and will result in considerable reduction of weight and drag.

One drawback with chemical cooling has been its apparent tendency toward excessively high oil temperatures. It seems reasonable, therefore, to design the engine to operate at high oil temperatures, developing an oil with sufficient body at high temperatures to be satisfactory.



Launching of the Improved-Dushy S.S. motor in the 1911 Biplane Dushy Boat.

In addition to these features the use of a supercharger seems a reasonable method by which to obtain increased power at small expenditure of weight or drag. The use of the supercharger will in the case of low compression engines, at least, result in 20 to 30 per cent increased power. In addition, the number of carburetors may easily be reduced and the overall performance of the engine improved by means of the more uniform mixture delivered to the combustion chamber. The supercharger could be installed in place of the reduction gear normally used or, if the use of the latter seems desirable, in conjunction with it.

The most desirable engine for our 400 m.p.h. plane engine, therefore, to be of either a Diesel type or of a two stroke cycle type in the latter case with a supercharger. The engine should be of such section as to be included within a circle of diameter not greater than 36 in. or within an oval or other symmetrical section of equivalent area with the propeller axis at the approximate center of area. If a reduction gear is used it should be located in the nose of the engine in such manner as to facilitate streamlining the nose of the plane. The use of a supercharger will permit the omission of the intake stack or stacks and the elimination of the drag thereof.

Having chosen upon the type of engine the design of the fuselage may proceed. If the cylinder block or other engine accessories protrude beyond the line of the fuselage, fairings must be provided for such parts. These fairings should be as long as possible with no abrupt changes in section or contour. In designing fairings it will be advantageous to study the direction of the air stream close to the fuselage and in the slip stream. It would seem to be an advantage from the standpoint of reduced drag to have these fairings in line with the actual high speed air stream rather than a direct flow and aft plane. In any case the number of fairings should be reduced to a minimum and, where possible, fairings should be combined. For instance, if bearings for cylinder blocks are required, the plane may



The 1911 Biplane Dushy Boat with a geared propeller engine. The high speed of this plane was greater than that of the racing plane and the power required was considerably less for the same propeller reaction which made the plane almost invulnerable.

be so designed that the strut fairing may be included in the cylinder fairing.

The open cockpit, with the windshield, is a particularly serious source of drag. There seems to be no reason for the use of an open cockpit except the preparation of pilots against being completely enclosed. It should, however, be possible to design a windshield which may be adjusted in height so that at all times the cockpit could be completely enclosed while for landing and take-off it could be opened to permit upward vision to the pilot. Means to keep the windshield clean or its cleaning in flight should be provided. The shield should be of such nature as to open fully, in case of emergency, at the touch of a lever.

All recent racing experience has indicated the necessity for supplying fresh air to the pilot. A small duct should be carried well forward in plane, opening to the open air immediately behind the propeller. The air entering this duct should be led to a point in front of the pilot and so exhausted as to sweep past his head. The stream should be so baffled that there will be no sensation of a blast striking him in the face. Means for exhausting the air around the rear edge of the windshield should be provided. The head rest of the cockpit should be faired into the vertical fin, the lower portion of which should be carried sufficiently forward.

The effect of otherwise well designed structures may well be spoiled by the method in which struts and wires are attached to the structural members of the plane. These parts should be installed at angles as nearly as possible approaching a right angle in order to reduce the mutual aerodynamic interference between adjacent portions of the plane. The proper and incorrect methods of attaching drag struts are indicated in Figures 2 and 3. In the latter case, the drag caused by interference between the side of the engine cowling and the strut are now known to be more serious than if the lines of the cowling had been expedient to include that portion of the strut adjacent to the engine.

These considerations regarding the beneficial effect of properly matching cowling and fairing are, of course, fundamental but they are largely the sequence of what is frequently overlooked in the design of high

speed planes. They become increasingly important as the speed is increased and if neglected may easily set up drag reactions which are considerable as is completely counter to the beneficial result of increased power. As has already been stated, the reduction of drag may be considered to be the most important feature affecting future high speed planes.

When the fairing out and shape fixed by the power plant, tail lands and the necessary attachment to the floats the nose and shape of the latter may be considered. The wings may well be considered with the floats, since in an endeavor to keep the wings out of the slip stream, the wings must be attached to the floats.

The wings themselves bear considerable weight and, if properly constructed, should load themselves to use as a portion of the fuselage system. It seems desirable, therefore, that the stem first bringing the back into a wing structure so that, in effect, a portion of the wing is installed in the space between the two floats. The latter should be widely spaced, in order to increase the stability on the water. The outboard section of the wing would be of camber to extend from other side of the floats and with a curved dihedral to provide water clearance in case of a low landing with wing low.

The section of wing between the floats should be so located as to be practically clear of water with the plane at rest. The water tight section of the wing could then provide a reasonable reserve of reserve buoyancy as that exceptionally large loads, the small area of providing reserve buoyancy, would not be required.

From the normal seaplane are responsible for a considerable portion of the drag so that any attempt to reduce their size and unreasonably result in increased speed of the plane. Then to save the use of the floats is a function of the weight of the plane any reduction in structural weight must result in smaller floats. The reserve buoyancy, normally expressed as a percentage of the load water line displacement, also indicates size of the floats and hence the drag of the landing gear. For the seaplane the normal seaplane the reserve buoyancy is at least 50 per cent. For racing planes the reserve buoyancy is usually about 30 per cent and the figure was formerly considered to be the safe minimum. The writer, however, has observed a racing plane in which the reserve buoyancy was reduced to 32 per cent without, actually affecting the water performance. It seems possible, therefore, to reduce the reserve buoyancy of the floats not including the displacement of wings to approximately 20 per cent with reasonable expectation of satisfactory water performance.

The form of fin and wing structure suggested should, of course, only be adopted after thorough test in the wind tunnel and model basin. The shape of the

deck of the floats with the flaring of the wing roots and the tapered nature of the wing trailing. The shape of the bottom of the floats falls within the province of the model team and must be carefully developed to provide smooth bottom surface to support the weight of the plane while planing at and below take-off speeds. The shape of the bottom float suited for satisfactory water performance should be adopted even at the expense of some additional aerodynamic drag. In particular, any tendency to set up between the floats a "wash" or "slosh" of solid water breaking over the top



Fig. 4: The present method of attaching float struts. The inverted method is shown in Fig. 5.

of the section of wing between the floats, should be avoided. In order to avoid this defect it might be well to locate the center wing section well forward of the center of gravity of the plane, obtaining the proper balance by providing compensating counter-bank of the outer sections of the wing system.

The wings of the plane must probably be of all metal construction. The requirement that they be water tight or relatively so, will introduce restrictions, but not insurmountable difficulties in the way of construction. Then, too, the wings will probably be provided with surface radiators for cooling the engine. This radiator might well be located in the center section of the wing, using a very thin pipe of corrosion resisting and sufficiently stiff even in thin gauges to be capable of resisting impact with the water, as the skin of the radiator. A radiator so located will be especially efficient at taking speeds where it would be in the wash of spray set up by the propeller and floats. Cooling at taking speeds has always been a problem with racing planes due to the required long run at very low speeds when the air speed over the wing radiator is not sufficient to afford proper cooling. Location of the radiator in a position in which it comes in contact with the spray should extremely improve the problem.

The oil system world, it appears, lie similar to that used in the Supermarine S-6 since this system seems to have been very effective. It is probable that less radiator area will be required in the case of the proposed plane than was used in the S-6 because of the high color-temperature differential due to the use of higher oil temperatures. The amount of heat radiated at any given speed per unit area of surface radiator is directly proportional to the difference between the temperatures of the oil and the surrounding air.

All the original features suggested for a racing plane of the new format, the suggested shape and type of

engine, the improved fuselage contour and the relocation of the wings were designed to reduce the overall drag and hence increase the speed range. These changes will have only minor effect on the landing speed, which is largely dependent on the wing area. The effect of improvement in speed range is, therefore, more noticeable at the high speed.

However, clean design, greater speed range and increased power will be of no avail if the power is not sufficiently converted into thrust through the propeller. Having plane operation requires one of the propeller under two extreme conditions, maximum air speed and maximum air speed, which normally requires propellers of entirely different characteristics. This difference is, greatly accentuated as the speed range increases. The designer is now faced with the problem of providing a propeller with pitch sufficiently low to provide sufficient thrust at gateway speeds for take-off without undue sacrifice of high speed efficiency. The two demands cannot be entirely reconciled in the ordinary propeller. The day when further progress toward greater speed will be halted by the failure of propellers to operate sufficiently efficiently at both upper and lower extremes of the speed range is, apparently, if, indeed, it has not already arrived. It is at least probable that the disappointing record of the S-6, considering the power available in the Rallo-Rayor engine, was due to the inefficient propeller used. It was probably necessary to cut the pitch of the propeller at an angle sufficiently low to permit take-off, accepting as unavoidable the decreased efficiency at high speeds.

It seems desirable, therefore, that any future program should devote considerable study to the provision of a



Fig. 5: Inverted method of attaching float struts. The float is shown in its normal position, the side of the struts meeting and the struts lie more directly in the line of the wing.

Best of the racing were expanded on in Fig. 6.

propeller with highest possible efficiency at both extremes of the speed range. One solution of the problem would seem to lie in the provision of a variable pitch propeller. That such a device can be developed within the next two years seems certain, if funds from the most racing program are made available in sufficient quantities. Propellers of this type should, however, be used only after long continued whirling tests followed by service tests on the training racing plane mentioned elsewhere.

Consideration of the propeller system leads inevitably to discussion of the use of a reduction gear. The writer has always deplored the use of a reduction gear while recognizing its usefulness in increasing propeller efficiency. The reduction gear increases the effects of the torque reaction tending to increase the difficulty of water handling and also of air control. However, with a high speed engine the reduction gear cannot be dispensed with. It seems possible and is certainly desirable that the supercharger be built into the reduction gear. In fact, the writer considers that of the two mechanisms the supercharger is the more important and should be given first consideration.

AN ARTIST'S CONCEPTION of a plane designed in accordance with the ideas previously outlined is given in the sketch, Figure 6 and 7. This design is presented as a "mere idea" for all the difficulties which have been described as inherent in existing types of racing planes and as a matter of fact, no data from wind tunnel or model basin tests have, as the knowledge of the writer, been obtained of a plane embodying one of the distinctive features described. It is possible that difficulties, particularly in water performance, would become evident if such tests were made. The writer does, however, wish to mention the opinion expressed previously that to follow blindly the policy of obtaining increased speed by increase of power is to invite disaster and disappointment. Whereas approximately 3400 hp. would probably be required to attain a speed of 400 m.p.h. with the conventional type using gasoline if the relation between the speeds and power of the S-5 and S-6 is any criteria, only 2000 to 2200 hp. would, it is estimated, be sufficient to obtain the same high speed with a plane of the proposed type. The plane depicted would, it appears, be a solution, but by no means the only one, to some of the difficulties now faced by any organization sponsoring the construction of Schneider Trophy contenders of the near future.

In this connection it might be pointed out that foreign designers are not inactive in bringing out bizarre designs intended to obtain speed by reduction of overall drag without the use of extraordinarily high powers. During the past several years engineers from Germany have spoken of a floatless racing plane with photographs of what is, apparently, a similar plane or Italian design shown on the water, have appeared in aeronautical trade periodicals. It is thus apparent that the unconventional features proposed for the next American plane are by no means as extraordinary as some of the methods being considered by foreign competitors.

THE MATERIAL REQUIREMENTS for any program should take into account not only the actual racing planes but also the planes for proper instruction of pilots and for the obtaining of test and design data.

For each two planes entered in the race, one primary high speed instruction plane would seem to be desirable. For such purpose any of the more efficient conventional high speed or ordinary period planes, provided with wings of reduced area so as to permit high speed landings as well as extreme high speed, and with floats with



Figures 6 and 7: Sketches of an airplane embodying the features suggested by the author in this article.

center buoyancy of approximately 60 per cent would be suitable. These planes should be used for the original training of pilots inexperienced in flying extremely fast planes, in practicing turns without loss of speed, in fast landings and in water handling of such planes.

If unconventional features, such as have been discussed are built into the racing planes, such ideas should be tested as far as possible before they are adopted for the races. For this purpose, a would seem desirable to build what might be called the training race plane. This plane would be of the same general type as the actual races and should use the same type of engine. It should, however, have lower landing speed and lower speed range than the actual races. This plane, used as an intermediate training plane, should have characteristics intermediate between those of the primary high speed planes and the actual races. The use of a racing vessel in the plane would provide an excellent opportunity for training the power plant prior to the actual race or to lead to the completion of racing plane.

THE FIRST PLANE of the type considered the racing plane should be considered an experimental plane, should be built as the "dog ship," and after being tested for water performance at various locations, should be used to lead to confirm design calculations.

The actual racing plane or planes should then be completed, incorporating therein all the changes and improvements learned desirable as a result of the construction and use of the training race plane and of the experimental "dog" plane. It is considered that if this program is followed soundly the difficulties apparently inherent in the preliminary operation of racing planes will have been discovered in the two experimental planes so that the losses involving tests and experiments, which are the normal portions of those interested in obtaining satisfactory operation of racing planes, will be avoided and any possible delays in the preparation of planes largely eliminated.

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